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1

CFD based Investigations for the Design of Severe Service Control Valves used in Energy Systems

2 3 Taimoor Asim^{*1} Matthew Charlton ² Rakesh Mishra³ 4 ^{1,3} School of Computing & Engineering 5 University of Huddersfield, Queensgate, Huddersfield HD1 3DH, UK 6 ² Weir Valves & Controls UK Ltd, Britannia House, Elland HX5 9JR, UK 7 ¹t.asim@hud.ac.uk, ²Matthew.Charlton@weirgroup.com ³r.mishra@hud.ac.uk 8 9 Abstract 10 Multistage severe service control valves are extensively used in various energy systems, such 11 as oil & gas, nuclear etc. The primary purpose of such valves is to control the amount of fluid 12 13 flow passing through them under extreme pressure changes. As opposed to the conventional 14 valves (butterfly, gate etc.), control valves are often installed in energy systems with geometrically complex trims, comprising of various geometrical features, formed by a 15 complex arrangement of cylindrical arrays. The pressure within the trim varies in controlled 16 17 steps and hence, cavitation resistance can be embedded in the trim through improved design process for the trim for severe service applications in energy systems. The flow 18 characteristics within a control valve are quite complex, owing to complex geometrical 19 20 features inherent in such designs, which makes it extremely difficult to isolate and quantify contribution of these features on the flow characteristics. One of the most important design 21 parameters of such trims is the flow coefficient (also known as flow capacity) of the trim 22 23 which depends on the geometrical features of the trim. The design of valves for particular performance envelop within the energy systems depends on effects of complex trim 24 25 geometrical features on performance characteristics; hence, the focus of recent research is on 26 quantifying the hydrodynamic behaviour of severe service control valves, including the trims. This includes the estimation of the local flow capacity contributions of the geometrical 27 features of the trim through detailed numerical investigations. In this work, a tool has been 28 developed that can be used to predict the local contribution of geometrical features on the 29 flow coefficient of the trim. It is expected that this work will result in better performance of 30 31 the energy systems where these valves are used. 32 Keywords: Computational Fluid Dynamics (CFD), Severe Service, Control Valves, Flow 33 Capacity, Energy Systems 34

36 **1.0 Introduction**

35

Valves are an integral part of any piping network and are used in a variety of industries for 37 38 various process control applications. The design of valves is a specialist area and the performance of valves is integral to the performance of the energy systems. The severe 39 service control vales typically have very complex flow paths and it is necessary to have 40 understanding of flow characteristics through the complex pathways to eliminate undesirable 41 effects such as vibrations, noise and cavitation in energy systems. The designs of such valves 42 are carried out with the help of well-known standards but many times undesirable local flow 43 44 effects cannot be eliminated through such designs. The standards are continuously updated to incorporate state of the art knowledge into the design process through extensive experimental 45 and numerical research work carried out all over the world. Newer designs are continuously 46

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being developed for energy systems for which design methods illustrated in standards may be 1 only partially applicable. In such cases a thorough fluid dynamic analysis is necessary to 2 design such valves. The performance of the energy system depends on placement of the valve 3 in the loop and fittings around it. Kang et al [1] have carried out both experimental and 4 numerical investigations on the effects of using various pipe configurations/fittings, 5 downstream the control valve, on the flow capacity of the valve. L, T, Y and + type 6 7 configurations have been used in these investigations. It has been reported that with the use of such fittings, the pressure losses are around 10% more than with no fitting/s attached thus 8 affecting the valve's performance drastically. It has also been observed that the numerical 9 10 simulations over-predict the flow coefficient of the control valve by 3-5%, as compared to the experimental findings. The measurement accuracy for the valve's flow coefficient was 11 estimated to be $\pm 10\%$. Limited information regarding the numerical modelling has been 12 13 provided by the authors, hence, detailed commentary on the reasons for these variations is not possible. Furthermore, the flow capacity recorded is for the whole valve system (including its 14 components), and local variations and contributions to the flow coefficient by various 15 geometric features have not been discussed. Beune et al [2] have also carried out both 16 17 numerical and experimental investigations on the discharge capacity of high-pressure safety valves. Fluid-structure interaction based numerical techniques have been used to analyse the 18 performance of the valve. A cavitation model, based on Rayleigh-Plesset equation, has been 19 developed and implemented in the numerical solver, assuming no-slip velocity condition 20 21 between the phases. It has been demonstrated that the experimental results match well with the numerical results when the cavitation model is implemented. 22

23

Lin et al [3] carried out detailed numerical investigations on the drag, lift, moment and 24 discharge coefficients, and the hydrodynamic forces acting on a butterfly valve for various 25 Valve Opening Positions (VOPs). It has been shown that SST-ko turbulence model best 26 predicts the flow behaviour within the valve, along-with 2nd order upwind discretisation 27 schemes. It has been reported that the Computational Fluid Dynamics (CFD) based predicted 28 29 coefficients and forces are in close agreement with the experimental results. Yang et al [4] have carried out detailed numerical investigations on a stop valve, and have reported the 30 complex flow structure within the valves. Wake induced vibrations for various valve 31 geometries have been shown to affect the valve geometry differently, with the low frequency 32 properties of the fluctuating pressure source being the main source of vibrations, both within 33 the pipe and the valve. Limited information regarding the effects of geometrical features on 34 the performance of the valve has been reported. Furthermore, An et al [5] has reported almost 35 36 linear increase in the flow capacity of a control valve as the valve opening position increases. 37 The flow capacity recorded corresponds to the whole valve system (including its components). The information regarding individual contributions of these components to the 38 39 global flow coefficient of the valve system has not been discussed. Moreover, Grace et al [6] have developed a parametric equation to predict the flow capacity of choke valve trims, based 40 on upstream geometric parameters. However, this model is most effective when the trim 41 consists of only a small number of ports. Again, the effects of various valve system 42 components on its flow coefficient have not been discussed. Unlike the simplified flow 43 geometry mentioned in An et al [5], sever service valves have a fairly complex geometry. 44 The flow field inside such valves is largely unknown. The experimental studies [7] have been 45 used to predict global performance parameters, such as flow capacity, but interrelation 46 between the geometry and the local flow field is largely unknown. This may causes 47 48 uncertainty with regards to the performance of the valves in safety critical applications such 49 as energy systems. 50

Li et al [8] developed a transient CFD based model to predict the hydraulics of a rectangular 1 full open valve tray (trim). 3D two-phase flow of gas and liquid has been analysed to develop 2 a new correlation of liquid hold-up. Interphase momentum transfer term has also been 3 calculated. It has been reported that CFD can be used as an effective tool in the design and 4 analysis of industrial trays. Wu et al [9] carried out numerical investigations on the flow-5 pressure characteristics of a pressure control valve for automotive fuel supply system. It has 6 7 been shown that as the valve opening increases, the flow coefficient also increases. Detailed investigations on the effects of various valve system components, on its flow capacity, have 8 not been carried out. Qian et al [10] carried out numerical investigations on the dynamic flow 9 10 behaviour of a pilot-control globe valve. It has been reported that the internal flow field of the valve is quite complex. Forces and displacements of the valve core have been recorded and 11 analysed at a given operating condition. Valdes et al [11] presented a methodology for 12 development of reduced order models that can be used to estimate the fluid flow and the flow 13 forces in hydraulic valves, as a function of reduced number of critical dimensions and 14 material properties. The methodology developed is based on incompressible flow and makes 15 use of CFD simulations in order to determine the flow resistance coefficient. The developed 16 model can be used to determine the effects of varying geometry on valve's performance. 17 However, the primary limitation of the developed model is that it is applicable to only those 18 kinds of valves in which the flow fields are similar to the valve used by Valdes. 19 20

21 Srikanth et al [12] carried out numerical investigations on compressible flow in a typical puffer type chamber. It has been observed that the velocity vectors in the middle plane of the 22 23 chamber depict swirling flow characteristics, with turbulent eddies. Static pressure on the same plane has been noticed to be highly fluctuating indicating highly complex flow 24 charcateritics within internals of valves. Amirante et al [13-15] carried out a series of 25 26 numerical investigations on the flow forces acting on a hydraulic directional control valve. Investigations carried out at various flow rates indicate that the maximum flow force occurs 27 when the recirculation flow rate vanishes. Moreover, the peak value of the flow force 28 increases with increasing flow rate, but its position remains fixed. There are however 29 differences in the pressure fluctuations due to the geometrical effects. Lisowski et al [16-18] 30 carried out a series of numerical investigations on the flow characteristics of a proportional 31 flow control valve. CFD based predictions on the pressure losses within the valve have been 32 shown to be within $\pm 5\%$ band as compared to the experimentally obtained data. CFD based 33 predictions have been used to generate new design features for better hydraulic efficiency of 34 the valves. 35

36

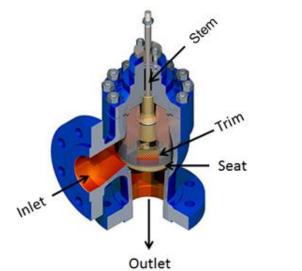
37 Critical analyses of the published literature regarding the flow behaviour and the flow capacity analysis of severe service control valves reveal that these analyses are carried out 38 mostly on the global performance parameters of the valve, such as the flow capacity of the 39 40 control valve system (valve and its components together). However, a better understanding of the local flow phenomena and local flow capacity of the different components of the control 41 valve system is extremely important in order to ensure better performance characteristics 42 during routine as well as safety critical applications in energy systems. In case of severe 43 service control valves, one of the most important components of the system is the trim. In the 44 present study, local flow field analyses within a trim has been carried out at various control 45 valve opening positions, in order to estimate the contribution of various geometrical features 46 of the trim on its local flow capacity. Furthermore, the contributions from other valve 47 48 components, such as the valve body and the seat, have also been enumerated. A combination of experimental and numerical investigations has been carried out to achieve this. In the next 49 section, the details of the methodology to calculate the flow capacity (global) of the 50

- components of the control valve system (valve, seat and trim) have been discussed, extending 1 it further to calculate the local flow capacity within the trim. 2
- 3

Flow capacity of a severe service control valve system 2.0

4 A control valve system comprises of three main components through which flow takes place 5 i.e. the valve body, the seat and the trim, as shown in figure 1 [19]. The control valve systems 6 are installed within severe service (high differential pressure) pipelines. The flow enters the 7 valve system via the inlet section of the system. The flow then enters the trim, which in the 8 9 present study, consists of stacks/layers (called disks) of staggered cylindrical columns, offering resistance to the flow. Hence, the fluid pressure drops in steps. Upon exiting the 10 trim, the flow enters the outlet section of the valve system, from where it propagates to the 11 outlet duct/pipe. The amount of flow passing through the control valve system is controlled 12 by an actuator. The actuator is connected to the stem which moves up and down the central 13 void section of the trim in order to open or close the control valve system to a prescribed 14

valve opening position. 15



16 17 18

Figure 1 Components of a Severe Service Control Valve [19]

19 The design of a control valve system is dependent on the requirements of the flow coefficient of the system, which can be computed as [20]: 20

21 22

 $Cv_{Control-Valve-System} = \frac{1}{\sqrt{\left(\frac{1}{Cv_{Valve Body}}^{2}\right) + \left(\frac{1}{Cv_{Seat}}^{2}\right) + \left(\frac{1}{Cv_{Trim}}^{2}\right)}}$ (1)

23

The structure of equation (1) is indicative of flow in series along the components of the 24

control valve system. The methodology to calculate the flow capacity of the severe service 25

control valve system has been extensively reported in the British Standard EN 60534-2-3 and 26 International Electrotechnical Commission 60534-2-3. These standards describe the sizing 27 equation for the non-choked, incompressible fluid flow in a severe service control valve as 28 29 [20-21]:

30

31
$$Q = N_1 F_R F_P Cv_{Control-Valve-System} \sqrt{\frac{\Delta P}{\rho_o}}$$
(2)

1 where Q is the volumetric flow rate of the fluid passing through the control valve system, N_1

2 is a numerical constant, F_R is the Reynolds number factor, F_P is piping geometry factor,

3 $Cv_{Control-Valve-System}$ is the flow capacity of the system, ΔP is the differential pressure across the

4 control valve system, ρ is the density of the fluid flowing in the control valve system and ρ_0 is

the density of water. The value of N_1 depends on the units used to compute equation (1). If

- 6 the volumetric flow rate is measured in m^3/hr and the differential pressure is measured in kPa, 7 the value of N₁ is 0.0865 [7]. The value of F_R depends on whether the flow within the valve is
- laminar or turbulent; for turbulent flows, its value is 1. The value of F_P depends on whether
- 9 any pipe fittings (such as a reducer, expander etc.) is attached to the valve. In case there are
- 10 no pipe fittings attached to the value, the value of F_P is 1. ρ/ρ_0 is the specific gravity of the
- 11 fluid, and for flow of water within the valve, its value is 1.
- 12

13 It has been mentioned in [21] that with the exception of valves with very small values of

14 Cv_{Control-Valve-System}, turbulent flow will always exist. It has been observed, while conducting

experiments in the present study, that $Cv_{Control-Valve-System}$ values are not very small, and hence

16 turbulent flow assumption seems reasonable, i.e. $F_R=1$. Furthermore, there has been no pipe

17 fitting used with the valve considered in the present study, hence the value of F_P is 1. Based

18 on the units used for Q and ΔP in the present study (m³/hr and kPa), and the working fluid

19 (water), equation (2) can be re-written as [20]:

20

$$Cv_{Control-Valve-System} = \frac{11.56 \text{ Q}}{\sqrt{\Delta P}}$$
 (3)

21 22

23 It can be noticed from equation (3) that the flow capacity of the control valve system is

24 directly proportional to the volumetric flow rate through the system and inversely

25 proportional to the square root of the differential pressure across the system. These values can

be measured both experimentally and numerically. It is noteworthy at this point that equation(3) is valid only for Newtonian fluids and for non-vaporizing conditions.

27 28

The flow capacity of the valve body and the seat can be computed, for the valve considered inthe present study, as [21]:

31 32

$$Cv_{Valve-Body} = k_1 \left(\frac{D_{Valve}}{D_{Seat}}\right)^2$$
 (4)

33 and,

$$Cv_{Seat} = k_2 \left(\frac{D_{Valve}}{D_{Seat}}\right)^2$$
 (5)

34 35

36 where k_1 and k_2 are coefficients that depend on the geometry of the valve and the seat.

Equations (4-5) have been developed based on CFD analyses of flow though these

38 components. As the flow field through these components is reasonably simple, the focus of

further investigations is towards the flow distribution within the trim. The flow coefficient of the trim (Cv_{Trim}) in equation (1) can then be computed as:

41

42

$$Cv_{Trim} = \frac{1}{\sqrt{\left(\frac{1}{\left(\frac{11.56 \text{ Q}}{\sqrt{\Delta P}}\right)^2}\right) - \left(\frac{1}{\left(k_1 \left(\frac{D_{Valve}}{D_{Seat}}\right)^2\right)^2}\right) - \left(\frac{1}{\left(k_2 \left(\frac{D_{Valve}}{D_{Seat}}\right)^2\right)^2}\right)}}$$
(6)

1 It should be noted that Cv_{Trim} in equation (6) is the global flow capacity of the trim i.e. across 2 the whole trim, which can be determined both experimentally and numerically.

3

4 It can be seen in equations (4-5) that the flow capacities of the valve body and the seat

5 depend on their geometrical features, which is also true for the trim. However, the

6 interdependence of trim's local flow capacity and its geometrical features cannot be

7 established using conventional experimental methods. Hence, in the present study, this

8 interdependence has been established with the use of CFD based techniques. Firstly, the

9 global flow capacity of the trim has been measured experimentally. This is then compared

against the CFD predictions of the same. Then, in-depth analysis of the local flow capacity of the trim, and various geometrical features of the trim, and at various valve opening positions,

the trim, and various geometrical features of the trim, and at various valve openinhas been carried out.

13

14 **3.0.** Estimation of global flow capacity of the trim

15 In order to determine Cv_{Trim} in equation (6), detailed experimental investigations have been 16 carried out in the present study. The flow capacity of the valve and the seat are known (based 17 on their diameters), and hence only $Cv_{Control-Valve-Svstem}$ needs to be computed experimentally

- 18 to determine Cv_{Trim} .
- 19

In accordance with BS EN 60534-2-5 [22], the test setup has been constructed, comprising of two straight lengths of pipe, connected to the ends of the valve, as shown in figure 2. The upstream pipe is 20 times longer than the nominal diameter of the pipe (d) while the upstream pressure tapings are attached at a distance of 2 * nominal diameter of the pipe, from the inlet of the valve. The downstream pipe is 7 times longer than the nominal diameter of the pipe

while the downstream pressure tapings are attached at a distance of 6 * nominal diameter of

the pipe, from the outlet of the valve. The nominal diameter of the pipeline is 100mm.

27

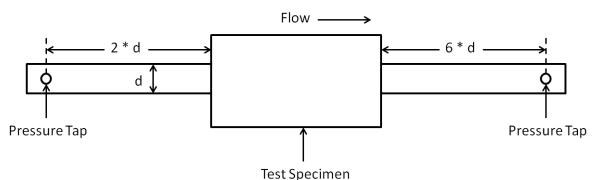




Figure 2 Dimensions of Upstream and Downstream sections

30

Clean mains water is supplied to the centrifugal pump from the water storage tank, once the

upstream ball valve is opened. A turbine flow meter is positioned downstream of the pump,

and upstream of the test valve, to monitor flow rate. The pressure taps of the pipeline are

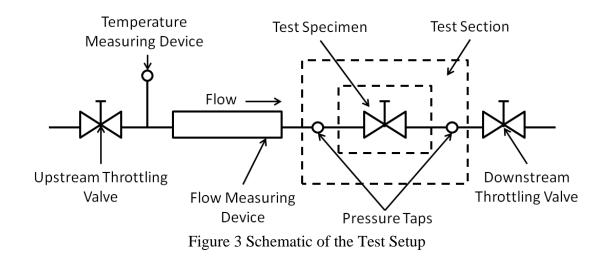
connected to a compact liquid differential pressure transducer in order to record the pressuredrop between the upstream and downstream pressure tap locations. The differential pressure

drop between the upstream and downstream pressure tap locations. The differential transducer measures differential pressure of upto 2.5bar, with an accuracy of $\pm 0.5\%$

37 (IEC60770). The output signal of the transducer is transmitted over a linear range from 4 to

20mA, which is converted into 0 to 10V using an AC-DC converter. The schematic of the test

setup is shown in figure 3.

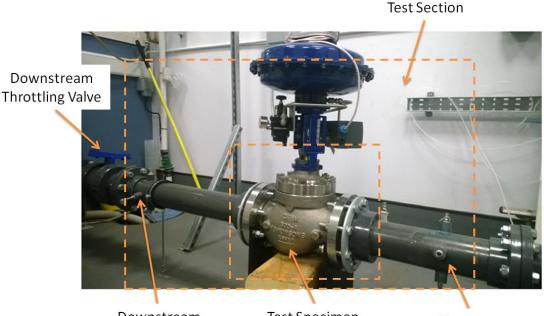


- 1 2
- 3

Figures 4 depict the installed flow loop setup for the capacity testing of the valve used in the 4

5 present study. The actuator sitting on the top of the valve is connected to a loop calibrator,

- which is further connected to a compressed air supply maintained at 4bar gauge. The loop 6
- calibrator is used to control the valve opening position. The loop calibrator has built-in 24V 7
- 8 DC power supply and measures 0 - 24mA DC current, with an accuracy of 0.01%. It can be
- further seen in figure 4 that four pressure tapings have been connected at both upstream and 9
- 10 downstream locations. These pressure tapings measure the average static gauge pressure at the specified location.
- 11
- 12



Downstream **Pressure Tap**

Test Specimen

Upstream **Pressure Tap**

- 13 14
- 15

Test runs have been carried out as per test procedure VT-QC-SP503 [21]. The main objective 16

Figure 4 Flow Loop Setup

- of the test programme is to determine the flow capacity of the control valve system in 17
- equation (3), from which the flow capacity of the trim can be computed as per equation (6). 18
- The tests have been conducted at valve opening positions of 100% (fully open), 80%, 60%, 19
- 40%, 20% and 10% in order to cover a wide range of operation of the control valve. 20
- 21

1 The experimental data for the aforementioned test runs is tabulated in table 1. Volumetric

2 flow rate (Q) in ltrs/min and the pressure drop across the valve (ΔP) in volts are recorded at

3 various valve opening positions. The flow rate and pressure drop across the valve are then

4 computed in m^3 /hr and kPa units, from which $Cv_{Control-Valve-System}$ is calculated using equation

5 (3). Using known values of $Cv_{Valve-Body}$ and Cv_{Seat} , Cv_{Trim} in equation (6) is computed.

6 7

Table 1 Experimental data for Cv_{Trim} VOP Cv_{Trim} Q ΔP Cv_{Control-Valve-System} Cv_{Valve-Body} Cv_{Seat} m⁷ m⁷ m⁷ m⁷ (m^3/hr) (%) (kPa) kg kg kg kg 100 51.8 342.84 32.3 301.6 65.0 37.5 80 45.9 354.48 28.2 301.6 65.0 31.4 60 36.6 371.28 22.0 301.6 23.4 65.0 40 25.6 378.36 15.2 301.6 65.0 15.7 20 11.1 373.44 6.6 301.6 65.0 6.7 375.00 10 7.1 4.2 301.6 65.0 4.2

8

It can be seen that at 10% valve opening position, the volumetric flow rate is $7.1 \text{m}^3/\text{hr}$, and 9 10 the differential pressure across the valve is 375kPa, hence, Cv_{Trim} works out to be 4.2. As the VOP increases to 20%, volumetric flow rate and Cv_{Trim} increase to 11.1m³/hr (56% increase) 11 12 and 6.7 (59.52% increase) respectively. Further opening the valve to 40% increases the 13 volumetric flow rate and the flow capacity of the trim by 130% and 134% respectively, as compared 10% opening values. Comparing the data between VOPs of 60% and 100% reveals 14 15 that volumetric flow rate and the flow capacity of the trim increases by 12.85% and 19.43% respectively from VOP of 60% to 100%. It can also be noticed that there are marginal 16 variations within the differential pressure across the valve at all VOPs. Hence, it can be 17 18 concluded that as the valve opening position increases, the volumetric flow rate across the valve increases, increasing the flow capacity of the trim. This can be further visualised in 19 20 figure 5.

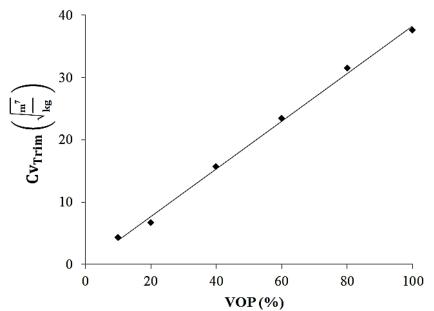




Figure 5 Variations of Cv_{Trim} with valve opening positions

1 The experimental findings presented here are in-line with the findings in other experimental

- 2 studies carried out by various investigators [23-24]. However, the primary limitation with the
- 3 experimental results is that they provide information about global Cv_{Trim} only, and not on the
- 4 quantitative effects of local geometrical features and their contribution towards overall Cv.
- This limits the information and the necessary knowledge to be able to optimise the valve
 geometry for better local as well as global performance characteristics. Computing the local
- flow capacity within the trim is an intricate task, which is accomplished in the present study
- by utilising the advanced Computational Fluid Dynamics based numerical techniques. With
- 9 the recent advancements in computational power, it has become possible to analyse the flow
- 10 behaviour within very complex geometries (like the one considered in the present study) with
- reasonable accuracy [25-28]. Hence, the following section/s present the details of the CFD
 modelling approach employed to locally analyse the capacity of the trim, and quantify local
- 13 contributions.
- 14

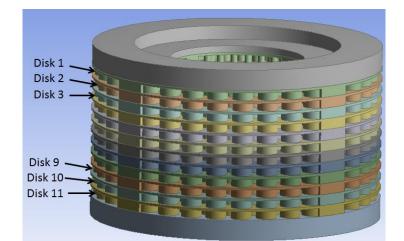
15 **4.0 Local flow capacity of the trim**

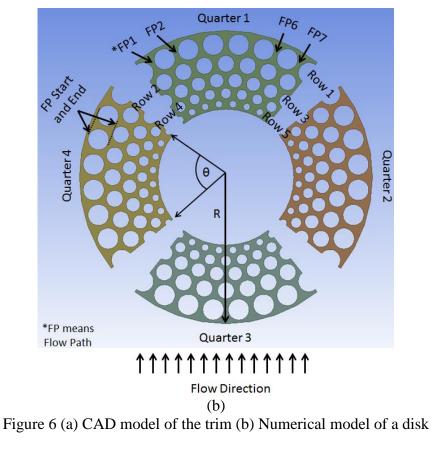
A trim used in severe service control valves has complex geometrical features. These features interact with the flow field in a complex manner. An attempt has been made here to develop an analytical model to quantify the local flow capacity of the trim based on these geometrical features.

20

A valve trim is a geometrically complex structure, consisting of stacks of disks, where each disk consists of a number of rows (formed by the cylindrical arrays) and flow paths. The trim used in the present study for analysis is shown in figure 6, where figure 6(a) shows the CAD model of the trim, while figure 6(b) depicts the numerical model of a single disk within the

- trim. It can be clearly seen that the trim under consideration comprises of 11 disks, where
- each disk comprises of 5 rows. Each row then comprises of multiple flow paths, which areformed between the cylindrical arrays of the same row. Each row has different number of
- flow paths i.e. rows 1, 3 and 5 have 7 flow paths, whereas, rows 2 and 4 have 8 flow paths. It
- is also noteworthy that the end flow paths (FP1 and FP7/8) have different shapes for different
- rows due to the arrangement of the cylindrical arrays in that row. Hence, it is expected that
- 31 the end flow paths for different rows will exhibit different flow behaviour.
- 32





In order to determine the local flow capacity within a trim, Cv needs to be calculated in disks,
rows and flow paths. As a trim consists of a number of disks, where the flow enters each disk
simultaneously, the flow capacity of a trim can be computed, in terms of disks, as:

$$Cv_{Trim} = \sum_{1}^{i} Cv_{Disk i}$$
(7)

where i is the total number of disks in the trim, which in the present study is 11. The structureof equation (7) is typical of the flow in disks in parallel.

14 Each disk consists of a number of rows, formed by the cylindrical arrays. As the flow

15 propagates through the rows predominantly in radial direction, the flow capacity of the disk,

16 in terms of rows, can be represented as:

$$Cv_{Disk} = \frac{1}{\sqrt{\sum_{1}^{j} \left(\frac{1}{Cv_{Rowj}^{2}}\right)}}$$
(8)

19

1 2

3

4

9 10

13

17 18

where j is the total number of rows, which in the present study is 5. It can be noticed that thestructure of equation (8) is that of flow in series.

22

Each row consists of a number of flow paths through which the flow can take place. The

24 number of flow paths depends on the number of cylinders within that particular row. As the

25 flow enters each flow path of a particular row simultaneously, the flow within flow paths is

treated similar to the flow in parallel, and hence, the flow capacity of a row can be

27 represented, in terms of flow paths of that row, as:

1

2

3

$$Cv_{Row} = \sum_{1}^{k} Cv_{Flow-Path k}$$
(9)

where k is the total number of flow paths within that particular row. k is variable in the 4

5 present study. Its value is 7 for odd number of rows, and 8 for even number of rows.

Combining equations (7-9), Cv_{Trim} can be expressed as: 6

7 8

$$Cv_{Trim} = \sum_{1}^{i} \frac{1}{\sqrt{\sum_{1}^{j} \left(\frac{1}{\left(\sum_{1}^{k} Cv_{Flow-Path}\right)^{2}}\right)}}$$
(10)

9

 $Cv_{Flow-Path}$ in equation (10) can be computed using equation (3), however, the volumetric flow 10 rate (Q) and the differential pressure (ΔP) in that case will be across the flow path, and not the 11 whole trim. O and ΔP across individual flow paths cannot normally be measured 12 experimentally; hence, the flow capacity of the trim in equation (10) has been computed 13

numerically. CFD based analysis have been carried out in the present study to analyse the 14 15 flow distribution within the different sections of the trim, which affects the local flow

capacity of the trim. Moreover, the effect of VOP on flow distribution has also been analysed 16

in detail. 17

18

5.0. 19 Numerical modelling of the control valve

The three dimensional numerical model of the control valve is shown in figure 7(a), where 20 the flow direction is from right to left. The inlet and outlet of the flow domain have been 21 modelled according to the industrial standards, as discussed before (pressure tapping 22 locations). The inlet and outlet pipe sections have been modelled in a different manner than 23 the valve itself, in order to employ different meshing techniques and sizes to the valve and 24 trim surfaces.

25

26

27 The concept of hybrid meshing has been used for meshing the flow domain. The inlet and outlet pipe sections have been meshed using hexahedral elements, while the valve and trim 28

29 have been meshed with tetrahedral elements. The inlet and outlet pipe sections have been

prescribed with a constant mesh element size of 3mm. In order to establish that the results 30

predicted by the numerical solver are independent of the mesh sizing within the test section, 31

32 three levels of mesh sizing have been used in the present study. Table 2 summarises the mesh

sizing within the test section, and also presents the results for mesh independence testing. It 33

can be seen that the test section has been meshed with minimum and maximum sizing of 0.3. 34

35 0.5 and 0.7mm, and 3, 5 and 7mm respectively. The mass flow rate predictions suggest that

the mesh with the minimum and maximum sizing of 0.3mm and 3mm respectively is capable 36 of predicting the flow variables within the test section with reasonable accuracy, and hence 37

38 has been chosen for further analysis. This mesh of the flow domain is shown in figure 7.

39

40 Three dimensional Navier-Stokes equations, along-with the continuity equation, have been

41 numerically solved in an iterative manner for the turbulent flow of water within the flow

domain. As far as the turbulent flow is concerned, two equation Shear Stress Transport (SST) 42

k-ω model has been chosen for turbulence modelling. The primary reason behind choosing 43

SST k-ω model is its superiority in accurately modelling the severe velocity gradients, which 44

are expected to occur within the trims due to complex flow path changes [29-32]. The SST k-45

ω model includes a blending function for near-wall treatment. It further has the definition of 46

1 the turbulent viscosity which is modified to account for the transport of the turbulent shear

2 stress. These features make the SST $k-\omega$ model more accurate and reliable for a wider range

3 of flows. Other modifications include the addition of a cross-diffusion term in the ω equation

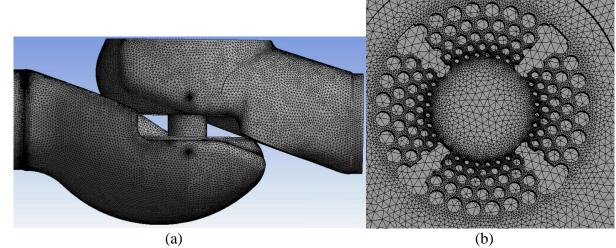
and a blending function to ensure that the model equations behave appropriately in both the near-wall and far-field zones. Further details of SST k- ω model can be found in any

6 turbulence modelling text book [33-35].

7 8

Table 2 Summary of mesh sizing and mesh independence testing			
Parameters	Level 1	Level 2	Level 3
Minimum size in the test section (mm)	0.3	0.5	0.7
Maximum size in the test section (mm)	3	5	7
Total number of mesh elements (millions)	5.2	3.8	2.2
Mass flow rate across the control valve system (kg/sec)	14.74	14.25	13.17
Difference in mass flow rate w.r.t. Level 1 results (%)		3.32	10.65

9



10 11 12

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Figure 7 Mesh in (a) the valve (b) the trim

In order to capture the complex flow phenomena associated with the control valve, especially 14 the resolution of the boundary layer flow, mesh layers have been concentrated in the near-15 wall region, where the boundary layer forms. As it has already been discussed that SST k- ω 16 turbulence modelled has been considered in the present study for modelling turbulence in the 17 flow, the mesh layers have been placed at strategic locations away from the walls. These 18 19 locations are based on the fact the SST k- ω turbulence model models the viscous sub-layer (i.e. y+ values upto \sim 5) and the buffer layer (i.e. y+ values from \sim 5 upto \sim 12). However, SST 20 k- ω turbulence model resolves the flow in the log-law region (i.e. y+ values from ~12 upto 21 \sim 300). Hence, the mesh layers are concentrated in the log-law region.

22 23

24 The inlet and outlet boundaries of the flow domain have been specified with total and static 25 gauge pressures respectively. The differential pressure across the control valve has been kept

the same as while performing the experiments in the laboratory, which ranges from 341.3kPa

at 100% VOP to 375kPa at 10% VOP. The walls within the flow domain have been modelled
as stationary walls with no-slip condition.

4 5.1. Verification of CFD results

5 In order to ascertain the accuracy of the numerical modelling and the solver settings used in 6 the present study, CFD predicted results need to be verified against the experimental findings. 7 In the present study, this has been carried out on Cv_{Trim} , tabulated in table 3. It can be seen 8 that CFD predicted capacity of the trim matches closely with the experimentally calculated 9 Cv_{Trim} . The percentage difference in the two Cv_{Trims} , on average, is 1.6%, most part of which 10 is due to numerical convergence.

11

3

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Table 3 Benchmarking CFD results			
VOP (%)	Experimental Cv _{Trim}	CFD predicted Cv _{Trim}	Percentage difference in CFD predicted Cv _{Trim} w.r.t. Experimental Cv _{Trim}
(%)	$\left(\sqrt{\frac{\mathrm{m}^{7}}{\mathrm{kg}}}\right)$	$\left(\sqrt{\frac{m^7}{kg}}\right)$	(%)
100	37.52	36.29	-3.27
60	23.41	23.61	0.88
40	15.68	16.02	2.17
10	4.24	4.22	-0.33

¹³

14 After it has been established that the CFD predicted results are in close agreement with the

15 experimental findings, detailed qualitative and quantitative analyses on the flow behaviour

and the local variations in the capacity of the trim, at various valve opening positions, need tobe carried out to understand complex geometry-flow interaction.

18

19 Further establishing the superiority of the current numerical modelling approach, a

 $\label{eq:comparative study for the numerical prediction of CV_{Trim} has been carried out. Green et al$

21 [27-28] carried out CFD based analysis on the capacity testing of the same trim and the

control valve as considered in the present study. The main difference between the two studies

is the fact that Green et al considered only one quarter of a single disk for numerical

24 modelling, assuming that the capacity of each quarter of the trim, and each disk of the stack, 25 is the same. However, the CFD based predictions clearly showed significant over-prediction

of CV_{Trim} values (56.28 as compared to 36.29 in the current study). Hence, it can be

concluded that the numerical modelling approach used in the present study. Hence, it can be

in predicting both the trim's and the valve's capacity.

29

30 **6.0. Performance analysis of the trim**

In order to visualise the flow structure within the trim, figure 8 depicts the velocity vectors within the top disk of the trim at fully open valve position. The velocity vectors shown in the

figure corresponds to quarter 1 of the trim as shown in figure 6(b). The flow field

corresponding to only one quarter is shown here as it is expected that it will be similar in

35 other quarters of the trim as well. The flow direction through the trim is inwards i.e. through

row 1 to row 5, where row 1 corresponds to the largest sized cylinders and row 5 corresponds

to the smallest sized cylinders. It can be seen in the figure that as the flow passes through row

1 of the trim, it accelerates to a velocity magnitude of about 12.5m/sec. This increase is

39 expected as the flow area progressively decreases up to the middle section of the cylinders,

1 resulting in higher flow velocity. The flow velocity can then be seen to decrease to a value of

- 2 7.5m/sec, within the same flow path, because of increase in flow area, before entering row 2
- 3 of the trim. The flow through row 2 of the trim has same characteristics as noticed in row 1,
- 4 with the maximum flow velocity magnitude being 11.5m/sec, and the exit flow velocity being
- 5 7m/sec. The same features are observed in the flow through rows 3, 4 and 5. However, at the
- 6 exit of row 5, the flow features are completely different because of different geometrical
- 7 configurations next to row 5. The above mentioned non-uniformities in the velocity field
- 8 within the trim increases the hydrodynamic losses within the trim, which is discussed in more9 detail later [36].
- 9 c 10

After visualising the flow behaviour within the trim, flow parameters, such as static gauge 11 pressure and velocity magnitude, have been critically analysed for better understanding of the 12 13 complex nature of flow phenomena within the trim. These parameters uniquely represent the flow capacity through a flow passage and hence can be used later to establish effects of 14 various flow passages on overall flow capacity of the valve. Figure 9 depicts the static gauge 15 pressure variations within the quarter 1 of the top disk of the trim at 100% valve opening 16 position. It can be seen that the pressure is high at the entrance of the trim. The flow then 17 enters the flow path of the 1st row, where, due to area reduction, the static pressure decreases. 18 In the later part of the flow path, as the flow area increases, the static pressure gradually 19 recovers. The flow leaving the 1st row enters the flow path of the 2nd row, where the same 20 21 phenomenon occurs i.e. area and pressure reduction in the first half of the flow path and vice versa in the later half. The same phenomena repeat in all the rows of the trim, until the fluid 22 23 leaves the trim. This indicates that the pressure drop occur in a series of steps (equal to the number of rows) as the flow takes place through the trim. Furthermore, it can also been 24 25 observed that very low pressure regions exist on either sides of a cylinder (more evident in 26 row 5), which is typical of flow taking place over a circular cylinder. Hence, the possible locations within the trim that are more prone to cavitate are the reduced flow areas between 27 the cylinders, where the pressure can locally drop below the liquid vapour pressure of water 28 29 [37-40].

30

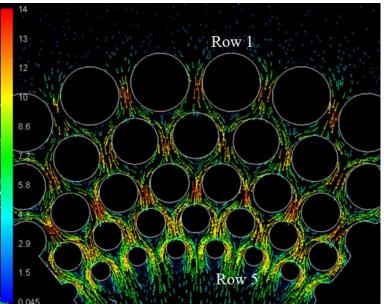


Figure 8 Velocity vectors within quarter 1 of the top disk of the trim at 100% valve opening position

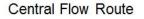
In order to further analyse the pressure variations with the trim, figure 10 depicts the local 1

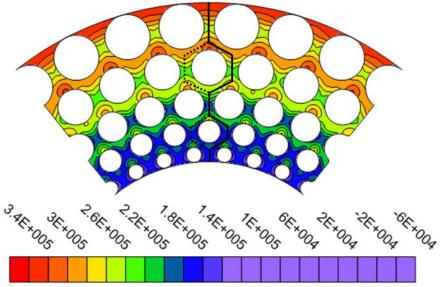
2 variations in static gauge pressure of the fluid as it passes through the central flow route of the top disk (shown in figure 9). The static pressure has been non-dimensionalised with the 3

dynamic pressure of the fluid within the trim, at the point where maximum flow velocity is 4

achieved. The x-axis corresponds to the radial dimension of the trim, where R is the outer 5

- radius of the trim and r is the local radial coordinate where the pressure has been recorded (as 6
- 7 shown in figure 6(b)). The vertical dotted lines represent the area between two rows i.e.
- inclined lines in figure 9. 8
- 9





10 11

Figure 9 Static gauge pressure (Pa) variations within the top disk of the trim at 100% valve 12 opening

13

14 The pressure variations corresponding to the top disk, shown in figure 10, depict that the pressure drops in a series of steps as the fluid flows through the trim. It can be clearly seen 15 that the non-dimensional static pressure at the entry of row 1 is 1.65. As the flow enters the 16 17 flow path of row 1 i.e. FP R1, due to variations in flow path's area, the pressure first decreases and then increases. At the exit of row 1, the non-dimensional static pressure 18 recovers to 1.5. Between the exit of row 1 and the entry of row 2, the non-dimensional static 19 20 pressure varies marginally. This is shown through two consecutive vertical dotted lines drawn in figure 10. The same flow phenomenon is seen to occur repetitively until the flow exits the 21 22 trim. As it has been identified that the pressure variations within flow paths occur due to variations in the flow areas within the flow paths, it is important to establish the interrelation 23 between area change and pressure variations. This has been achieved through a parameter 24 25 called Flow Area Ratio ($\xi_{n+1/n}$) that has been defined as:

26 27

28

$$\xi_{n+1/n} = \frac{\text{Flow Area}_{n+1}}{\text{Flow Area}_{n}}$$
(11)

where n is the row number. Hence, $\xi_{n+1/n}$ is the ratio of available flow areas between 29

consecutive rows, measured at the middle plane of the cylinders. $\xi_{2/1}$ i.e. effective flow area 30

in row 2 divided by the effective flow area in row 1, is 1.13, which means that the effective 31

- flow area in row 2 is 13% more as compared to row 1. Similarly, $\xi_{3/2}$ is 0.89, $\xi_{4/3}$ is 1.21 and 32
- $\xi_{5/4}$ is 1.26 respectively. It can be noticed that $\xi_{4/3}$ and $\xi_{5/4}$ are more than 1; however, $\xi_{3/2}$ is 33

- 1 less than 1. The effective flow area in row 3 is 11% less than in row 2, and 17% less than in
- 2 row 4. As the effective flow area in row 3 is less than that in rows 2 and 4, the flow
- 3 characteristics in the vicinity of row 3 will be different, resulting in higher hydrodynamic
- 4 losses.

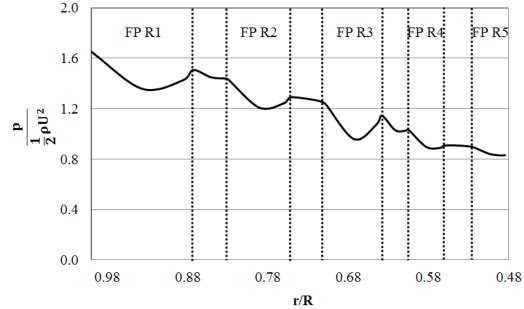
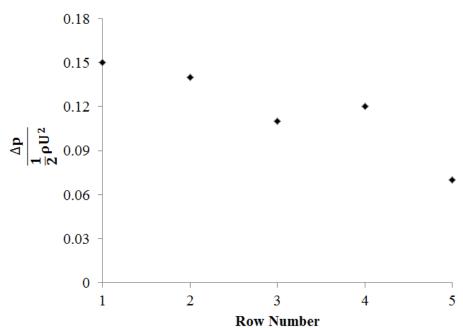


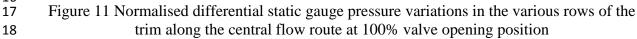
Figure 10 Normalised static gauge pressure variations along the central flow route within the top disk of the trim at 100% valve opening position

In order to establish the relationship between flow area ratio and the pressure variations
within the trim, non-dimensional differential static pressure, across the flow paths of the top
disk, have been plotted in figure 11. It can be seen that, for the top disk, the differential
pressure decreases from row 1 to row 2. The same trend is observed from row 2 to row 3,
however, from row 3 to row 4, the differential pressure increases, instead of the decreasing

- trend seen till row 3. From row 4 to row 5, the trend is similar to that observed upto row 3.
- 15







After quantitatively analysing the variations in pressure within the trim (for ΔP calculations in 1 equation (6)), variations in the flow velocity magnitude within guarter 1 of the top disk of the 2 trim at 100% valve opening position are depicted in figure 12. The volumetric flow rate 3 calculations in equation (6) are dependent on these variations, hence in order to calculate the 4 5 flow capacity of the trim, detailed quantitative analysis of the velocity variations is important, and is presented here. It can be clearly seen that the flow velocity magnitude increases in the 6 narrow passages formed between cylindrical arrays within a row. However, continuing from 7 previous discussion, the flow velocity magnitude is highest in the flow paths of the 3rd row, 8 compared to flow paths of other rows. The maximum flow velocity magnitude in each row is 9 12m/sec, 11.5m/sec, 12.7m/sec, 11m/sec and 9.9m/sec for rows 1 to 5 respectively. It is the 10 effective flow area ratios that are responsible for this behaviour as discussed before. The 11 information regarding the maximum flow velocity magnitude variations within the trim is 12 important for the design of the trim; in the stages where the maximum erosion rate 13 14 calculations are required. It is beneficial to keep the local flow velocity magnitude within certain limits to ensure minimal erosion and for reduction in hydrodynamic losses. 15

16

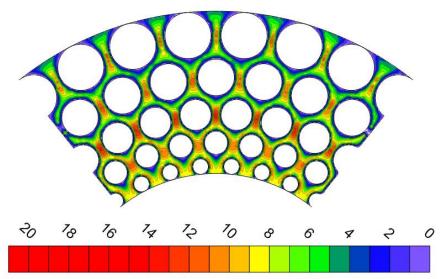
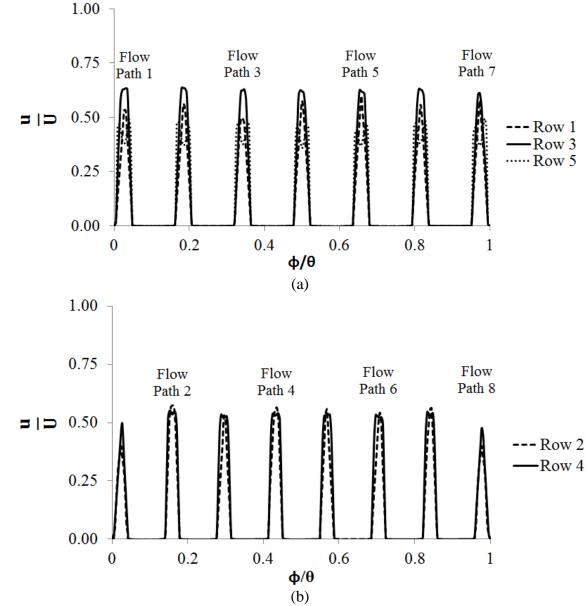


Figure 12 Flow velocity magnitude (m/sec) variations within the top disk of the trim at 100% valve opening position

20

21 In order to quantitatively analyse the flow velocity magnitude within the different flow paths of the trim, normalised velocity profiles have been drawn in each flow path, in figure 13. The 22 geometric details of these flow paths have been represented in terms of two parameters i.e. θ 23 and ϕ , where θ is the total circumferential dimension covered by a quarter of the disk (77° in 24 the present study), and ϕ is the local circumferential location (as shown in figure 6(b)). 25 Figure 13(a) depicts normalised flow velocity magnitude profiles within the flow paths of 1st, 26 3rd and 5th row, while figure 13(b) depicts normalised flow velocity magnitude profiles within 27 the flow paths of 2nd and 4th rows of the top disk. This differentiation is due to the fact that 28 there are different numbers of flow paths in different rows (rows 1, 3 and 5 have 7 flow paths, 29 while rows 2 and 4 have 8 flow paths each). Furthermore, the flow velocity magnitude has 30 been normalised with the maximum flow velocity within the trim, at any given valve opening 31 position. As the maximum flow velocity magnitude of 20m/sec has been recorded in case of 32 10% VOP, hence, the flow velocity magnitude profiles throughout this study have been 33 normalised against this value for effective comparison purposes. 34



1 2



Figure 13 Normalised flow velocity magnitude profiles within different flow paths of the top
disk at 100% VOP (a) for odd number of rows (b) for even number of rows

7

8 It can be seen in figures 13(a and b) that the flow velocity is maximum in the middle of the 9 flow paths because of the wall effects on either sides, complying with no-slip boundary 10 conditions specified to the solver. The maximum normalised flow velocity magnitudes in rows 1 to 5 are 0.6, 0.57, 0.63, 0.55 and 0.49 respectively. Hence, the global (in all the flow 11 paths together) maximum normalised flow velocity magnitude is observed in row 3 of the 12 trim, because of the relative flow area ratios. It can be seen that the local (in that particular 13 flow path) maximum normalised flow velocity magnitudes decrease from rows 1 and 2 by 14 15 5%, rows 3 to 4 by 12.7%, and rows 4 to 5 by 11%, whereas it increases from row 2 to 3 by 10.5%. Hence, row 3 is contributing the most towards the hydrodynamic losses and erosion 16 within the trim. It can be further seen that although the different velocity profiles are similar 17 to each other, there are slight variations in the 1st and last flow paths of rows 2 and 4, in 18 figure 13(b). The reduction in flow velocity in these flow paths is due to the geometrical 19 configuration of these flow paths. It can be seen in figure 6 that the end flow paths of these 20

two rows have a straight wall at one end each, whereas, in case of rows 1, 3 and 5, both walls
of the end flow paths are curved, formed by cylindrical arrays.

3

4 To estimate the contribution of various geometrical features to the global flow capacity of the trim, table 4 summarises the local flow capacities of different flow paths within the top disk 5 of the trim at 100% VOP. It can be seen that the local flow capacity remains almost constant 6 7 within the different flow paths of a particular row, although there are slight variations in the first and last flow paths of rows 2 and 4, due to wall effects as discussed earlier. It can be 8 seen that average local flow coefficients in rows 1 to 5 are 0.275, 0.253, 0.248, 0.255 and 9 10 0.348 respectively. Hence, the local flow coefficient decreases from row 1 to 2 by 8.2% and 11 row 2 to 3 by 2%, while it increases from row 3 to 4 by 3.1% and from row 4 to 5 by 36.5%. If the end flow paths of the 2nd and 4th rows are not considered, then the average local flow 12 coefficients of these two rows are 0.294 and 0.295 respectively. In that case, there will be 13 6.7% and 1.92% increase in the average local flow capacity from row 1 to row 2 and row 3 to 14 row 4 respectively. Hence, the estimation of local flow capacity contribution by the 15 geometrical features of the trim to its global flow capacity is very important. It can be further 16 seen that the 3rd row's contribution towards the global flow coefficient of the trim is the 17 lowest, while 5th row's contribution is the highest. This further suggests that the most and the 18 least hydrodynamic losses occur in rows 3 and 5 respectively. The local flow capacities of 19 flow paths, of a particular row, can be summed up to give the total local flow capacity of that 20 row, according to equation (9). Hence, the total local flow capacities of rows 1 to 5 are 1.65, 21 2.02, 1.73, 2.04 and 2.44 respectively. 22

23

	Row 1	Row 2	Row 3	Row 4	Row 5
Flow Path 1	0.277	0.127	0.249	0.131	0.350
Flow Path 2	0.277	0.297	0.250	0.304	0.349
Flow Path 3	0.271	0.298	0.249	0.290	0.346
Flow Path 4	0.273	0.301	0.246	0.300	0.344
Flow Path 5	0.280	0.285	0.244	0.290	0.338
Flow Path 6	0.278	0.287	0.244	0.296	0.346
Flow Path 7	0.272	0.295	0.251	0.291	0.365
Flow Path 8		0.130		0.139	

Table 4 Local flow capacity of different flow paths of the top disk of quarter 1 at 100% VOP

25

7.0. Local flow capacity variations in other quarters of the disk

It is important to evaluate the local flow capacities of the rows within each of the four
quarters of the disks in order to ascertain whether these quarters behave in the same manner

hydrodynamically or not. Hence, the total local flow capacities of all the rows (for all the
quarters of the top disk at 100% VOP) have been summarised in table 5. It can be clearly

quarters of the top disk at 100% VOP) have been summarised in table 5. It can be clearly seen that Cv_{Row} remains almost constant within different quarters of the disk. However, as

seen that Cv_{Row} remains annost constant within different quarters of the disk. However, as noticed earlier, Cv_{Row} for different rows is significantly different. Cv_{Row} of all the four

33 quarters are summed up to estimate the total local flow capacity of that particular row. Hence,

 Cv_{Row} have been computed to be 7.734, 8.097, 6.942, 8.168 and 9.768 for rows 1 to 5

respectively. It can be noticed that the total local flow capacity of the 3^{rd} row is minimum

- 1 amongst the different rows of the disk; hence row 3 is contributing the least towards the
- 2 global flow capacity of the trim. This is because row 3 is offering the most resistance to the
- 3 fluid flow within the trim, as compared to other rows, due to its effective flow area. The
- 4 increased resistance in row 3 increases the flow velocity within this row, hence increasing
 5 trim erosion.

Table 5 Local flow capacities of all the guarters for all rows of the top disk at 100% valve

- 6
- 7
- 8

opening position				
		Average ΔP across all flow paths within a row	Average Q of all flow paths within a row	$\mathrm{Cv}_{\mathrm{Row}}$
		(kPa)	(m ³ /hr)	$\left(\sqrt{\frac{m^7}{kg}}\right)$
	Quarter 1	36.59	0.144	1.928
Dow 1	Quarter 2	36.64	0.143	1.916
Row 1	Quarter 3	36.84	0.145	1.945
	Quarter 4	35.88	0.144	1.945
	Quarter 1	34.10	0.126	2.02
Row 2	Quarter 2	34.38	0.125	2.003
KOW Z	Quarter 3	33.83	0.128	2.064
	Quarter 4	33.73	0.125	2.01
	Quarter 1	45.80	0.145	1.733
Row 3	Quarter 2	45.33	0.144	1.731
	Quarter 3	46.32	0.146	1.741
	Quarter 4	45.31	0.144	1.737
	Quarter 1	33.65	0.127	2.041
Row 4	Quarter 2	33.24	0.127	2.043
	Quarter 3	34.41	0.128	2.042
	Quarter 4	33.35	0.126	2.042
Row 5	Quarter 1	23.67	0.146	2.438
	Quarter 2	23.34	0.145	2.441
	Quarter 3	24.07	0.148	2.443
	Quarter 4	23.27	0.145	2.446

9 10

8.0. Contribution of other disks of the trim to its local flow capacity

After analysing different flow paths, rows and quarters of the top disk at 100% VOP, other 11 disks of the trim, at the same VOP, need to be analysed in order to estimate their contribution 12 towards the global flow coefficient of the trim. For this purpose, the middle (5th) and bottom 13 (11th) disks have been analysed here. Figure 14 depicts the non-dimensional static pressure 14 ratio variations, w.r.t. the top disk, within the corresponding flow paths and rows of both the 15 middle and bottom disks of the trim It can be seen for the middle disk that the pressure ratio 16 is almost same as the top disk up to row 3. In row 3 of the middle disk, the variations in the 17 pressure are significantly higher than in the top disk. In the flow paths of rows 4 and 5 of the 18 middle disk, it can be noticed that the pressure, as compared to the top disk, is less. Similarly, 19 for the bottom disk, it is clear that there are significant differences in pressure with respect to 20 the top disk in the corresponding flow paths and rows of the trim. The pressure drops to much 21 lower values in case of the bottom disk, as compared to the top disk. 22

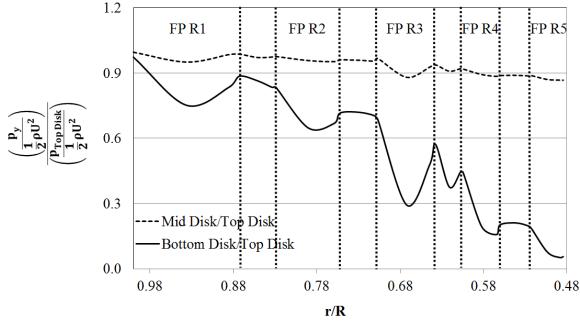


Figure 14 Normalised static gauge pressure ratio variations w.r.t. the top disk along the central flow route at 100% valve opening position

Further analysing the pressure variations within the middle and the bottom disks of the trim, 6 7 figure 15 depicts the variations in non-dimensional differential static pressure ratios, w.r.t. the top disk, across the flow paths of both the middle and the bottom disks of the trim. It can be 8 seen that the differential pressures in the middle disk are similar to the one for the top disk. 9 However, in case of the bottom disk, the differential pressures are significantly higher than 10 for the top disk. Moreover, it can be clearly seen that the differential pressure in row 3 of the 11 bottom disk is substantially more than in the 3rd row of the top disk. Another important point 12 13 to note over here for the bottom disk of the trim is that although it depicts lower pressure as compared to the top disk (see figure 14), the differential pressures across the flow paths of the 14 15 different rows are significantly higher. Hence, it is expected that more flow is taking place through the bottom disk of the trim as compared to the top and middle disks, which is 16 discussed in more detail later. 17

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19 Normalised flow velocity magnitude profiles for the different rows and flow paths of the middle and bottom disks have been plotted in figure 16, where figures 16(a and c) 20 corresponds to the profiles in rows 1, 3 and 5 of the middle and bottom disks respectively, 21 while figures 16(b and d) corresponds to the profiles in rows 2 and 4 of the middle and 22 bottom disks. It can be seen that, qualitatively, the trends are similar to the one observed in 23 case of the top disk. However, the maximum normalised flow velocity magnitudes in 24 25 different rows of the middle disk are 0.65, 0.62, 0.71, 0.6 and 0.53 respectively (from row 1 26 to row 5), which are 0%, 8.8%, 12.7%, 9.1% and 8.2% higher than for the top disk respectively. Similarly, the maximum normalised flow velocity magnitudes in different rows 27 of the bottom disk are 0.84, 0.8, 0.88, 0.77 and 0.67 respectively, which are 29.2%, 40.4%, 28 39.7%, 40% and 36.7% higher than for the top disk respectively. In all the disks, the 29 maximum normalised flow velocity magnitude is observed at the 3rd row, indicating 30 maximum erosion in this row, as compared to other rows of the disk. The reason for this 31 32 increase in the flow velocity magnitude is attributed to the amount of fluid flow passing through these disks. 33 34

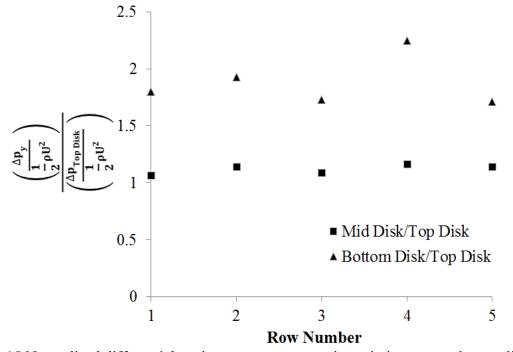
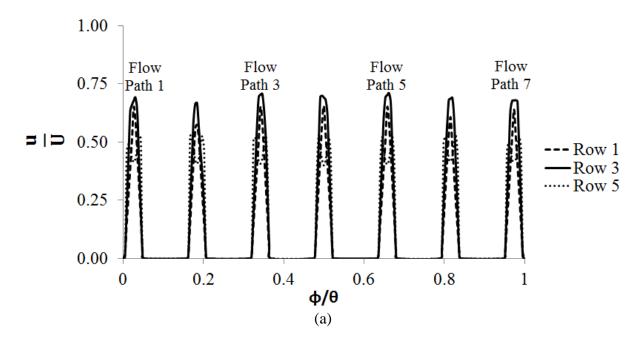


Figure 15 Normalised differential static gauge pressure ratio variations w.r.t. the top disk in various rows of the trim along the central flow route at 100% valve opening position





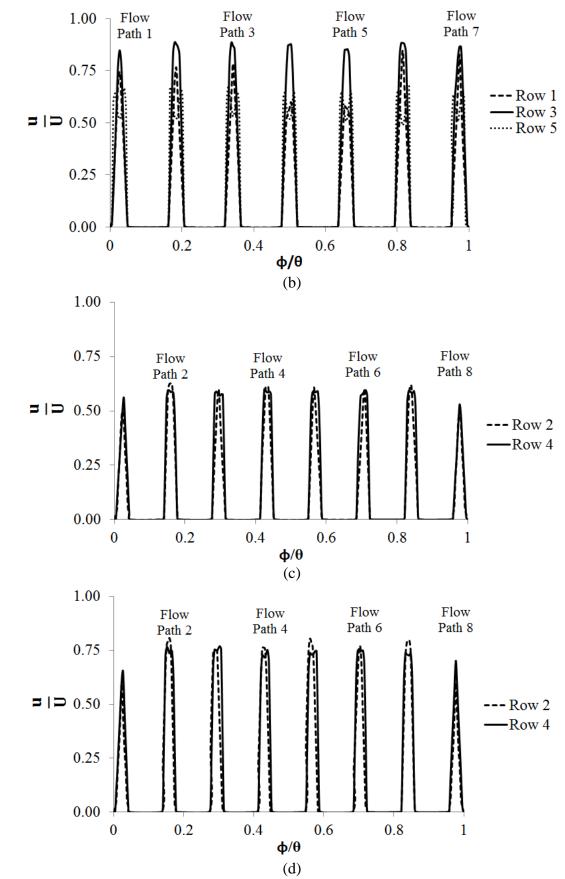


Figure 16 Normalised flow velocity magnitude profiles at 100% VOP within different flow paths of (a) middle disk for odd number of rows (b) bottom disk for odd number of rows (c) middle disk for even number of rows (d) bottom disk for even number of rows

1 Quantitatively analysing the reason for higher differential pressures and flow velocity in the middle and bottom disks of the trim, as compared to the top disk, table 6 summarises the 2 mass flow rate of water entering the trim through the various flow paths of the first 3 (outermost) row for the top, middle and bottom disks. It can be clearly seen that the amount 4 of fluid passing through the middle and bottom disks is higher than for the top disk. It is 5 noteworthy that, on average, the amount of flow taking place through the middle disk is 6.4% 6 7 higher than the top disk, while it is 40% higher for the bottom disk. It suggests that the middle disk offers less resistance to the flow of fluid as compared to the top row, while the 8 flow resistance is further reduced in case of the bottom disk. The effect this has on the flow 9 10 distribution within the trim is shown in terms of flow streamlines in figure 17. It can be seen

11 that the number of streamlines (amount of fluid flow) passing through the different disks of

12 the trim increases from the top to the bottom disks, due to the wall effects.

- 13
- 14 15

Table 6 Mass flow rate passing through the 1st row of top, middle and bottom disks of the trim at 100% VOP

	Ν	Aass Flow Rate (kg/se	ec)
	Top Disk	Middle Disk	Bottom Disk
Flow Path 1	0.0392	0.0408	0.0536
Flow Path 2	0.0407	0.0437	0.0557
Flow Path 3	0.0402	0.0418	0.0554
Flow Path 4	0.0410	0.0432	0.0579
Flow Path 5	0.0405	0.0433	0.0577
Flow Path 6	0.0395	0.0421	0.0569
Flow Path 7	0.0385	0.0427	0.0543
Average	0.0399	0.0425	0.0559



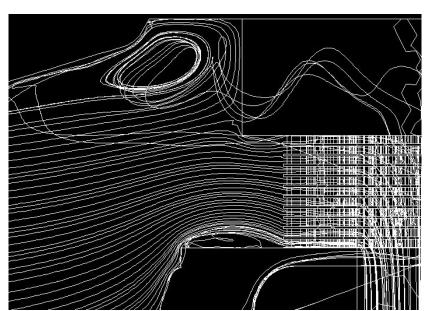


Figure 17 Streamlines passing through different disks of the trim at 100% VOP

9.0. Effect of valve opening position on the flow capacity of the trim

After carrying out detailed investigations in to the non-dimensional pressure and velocity 2 magnitude, and the local flow capacity of the different flow paths, rows, quarters and disks of 3 the trim, the next step is to estimate the effects of the valve opening position on the 4 5 contribution to local flow capacity of the trim by the various geometrical features. For this purpose, VOP of 60% and 10% have been considered in this section, and comparisons have 6 been made against 100% VOP wherever applicable. This analysis is important as the flow 7 distribution amongst the disks may change by changing the VOP, affecting the local flow 8 9 capacity of the disks and the global flow capacity of the trim. 10

11 Figure 18 depicts the non-dimensional static pressure ratio variations at various valve

12 opening positions, w.r.t. 100% VOP, within the corresponding flow paths and rows of the top

disk of the trim It can be seen for 60% that the pressure ratio is almost the same as for 100%

14 VOP in row 1 of the trim. From row 2 onwards, the variations in the pressure are

significantly higher than for 100% VOP; the pressure keeps on decreasing. It can also be

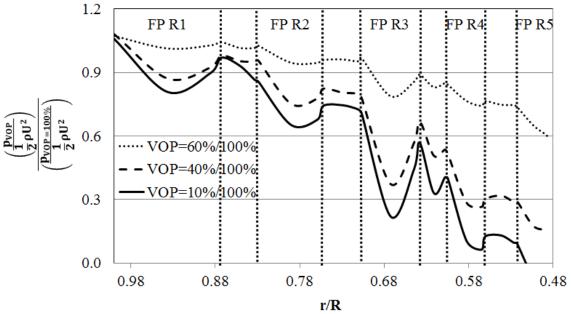
noticed that pressure decrease in row 3 of the disk is substantially more at 60% VOP as

17 compared to 100% VOP. Similarly, for 40% and 10% VOPs, it is clear that there are

significant differences in pressure with respect to 100% VOP in the corresponding flow paths

19 and rows of the trim. The pressure drops to much lower values, especially in row 3, in case of

- 20 both 40% and 10% VOPs, as compared to 100% VOP.
- 21



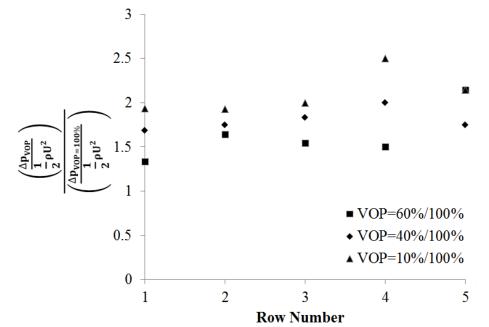
22 23

Figure 18 Normalised static gauge pressure ratio variations w.r.t. 100% VOP along the central flow route of the top disk of the trim

24 25

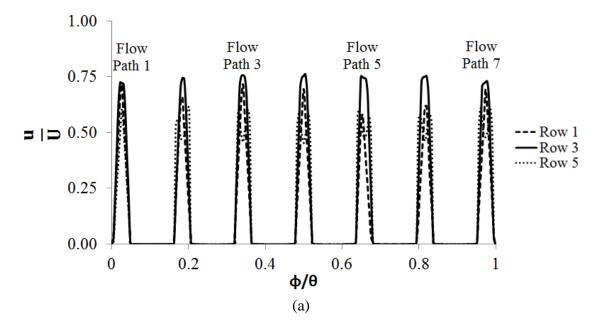
Further analysing the pressure variations at different VOPs, figure 19 depicts the variations in non-dimensional differential static pressure ratios, w.r.t. 100% VOP, across the flow paths of different rows of the trim. It can be seen that the differential pressures at all 60%, 40% and 10% VOPs are significantly higher than for at 100% VOP. It can also be seen that the differential pressure at 10% VOP is higher than at 40%, while the differential pressure at 40% VOP is higher than at 60% VOP. Furthermore, in the 3rd row of the disk, the differential pressure is substantially higher than at 100% VOP, indicating increased hydrodynamic losses

in row 3 of the trim.

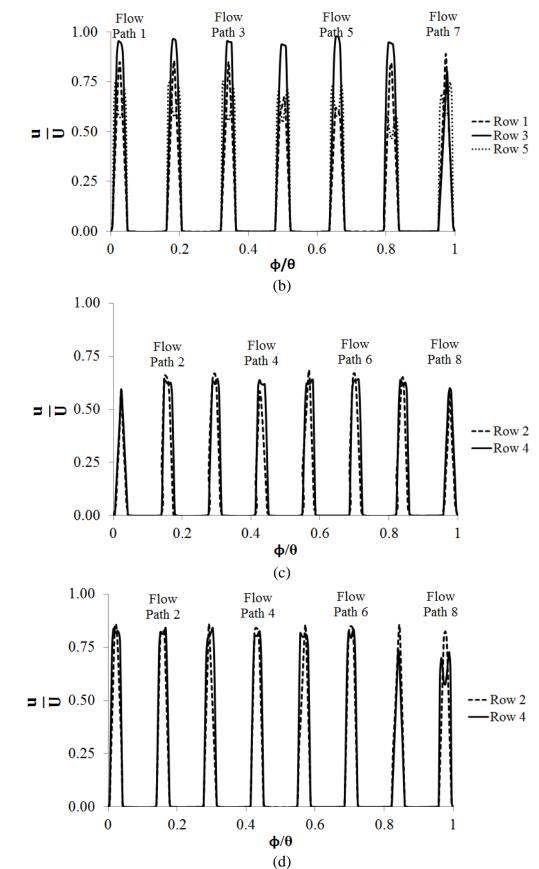


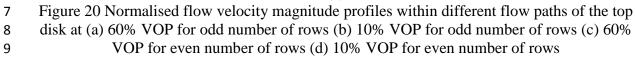
Row Number
 Figure 19 Normalised differential static gauge pressure ratio variations w.r.t. 100% VOP in
 various rows of the trim along the central flow route of the top disk of the trim

5 Normalised flow velocity magnitude profiles for the different rows and flow paths at 10%, 60% and 100% VOPs have been plotted in figure 16, where figures 20(a and c) corresponds 6 to the profiles in rows 1, 3 and 5 at 60% and 10% VOPs respectively, while figures 20(b and 7 d) corresponds to the profiles in rows 2 and 4 at 60% and 10% VOPs. The maximum 8 9 normalised flow velocity magnitudes in different rows at 60% and 10% VOPs are 0.71, 0.68, 10 0.76, 0.64, 0.61 and 0.65, 0.43, 0.66, 0.32 and 0.13 respectively. These are (for 60% VOP) 18.3%, 19.3%, 20.6%, 16.4% and 24.5% higher than for 100% VOP respectively. Similarly, 11 the maximum normalised flow velocity magnitudes in different rows at 10% VOP are 8.3% 12 higher, 24.6% lower, 4.8% higher, 41.8% lower and 73.5% lower respectively, as compared 13 14 to 100% VOP. At all VOPs, the maximum normalised flow velocity magnitude is observed at the 3rd row, indicating maximum erosion in this row. The reason for this increase in the flow 15 16 velocity magnitude is the amount of fluid flow passing through these rows at different VOPs. 17



18 19

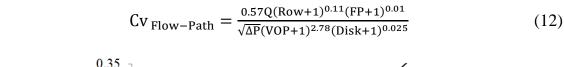


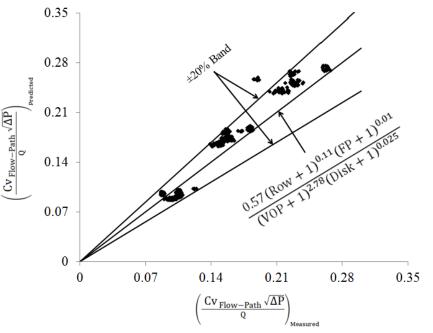


Based on the results presented in this study regarding the local flow capacity at a given 1 location within the trim (flow paths, rows, disks and at different valve opening positions), and 2 using advance statistical tools such as multiple variable regression analysis, a semi-empirical 3 local Cv prediction model i.e. Cv_{Flow-Path} in equation (10) has been developed and compared 4 against the measured local flow capacity of the trim. This prediction model is presented in 5 equation (12), which shows the local flow capacity of a flow path as a function of the valve 6 7 opening position (VOP), disk number of the trim, row number of the disk and flow path 8 number of the row. Volumetric flow rate (Q) and the differential pressure (ΔP) used to 9 develop this predicted model are taken from table 1, which are measured experimentally. It can be seen that as the valve opening position increases (from 0.1 to 1), local Cv decreases. 10 Substituting equation (12) in equation (10), as the VOP increases, Cv_{Trim} also increases. 11 Furthermore, it can be seen that as the disk number increases i.e. going from the bottom disk 12 to top disk, local Cv decreases. Moreover, increase in row number i.e. from outer (bigger 13 14 cylinders) to inner (smaller cylinders), local Cv increases. Equation (12) can be substituted in equation (10) to predict the global/total flow capacity of the trim, where all the inputs are 15 known in advance (Row, VOP, Disk and FP), or can be measured experimentally (Q and ΔP). 16 Figure 21 depicts a comparison between CFD measured and equation (12) predicted local 17 18 flow capacities. It can be seen that more than 90% of the data lies within $\pm 20\%$ band. Hence, equation (12) can be used to determine the local flow capacity of the trim with reasonable 19 20 accuracy.

- 21
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Figure 21 Comparison between measured and predicted normalised $\ensuremath{\text{Cv}_{\text{Flow-Paths}}}$

26

27 **10.0.** Conclusions

Severe service control valves are typically installed with geometrically complex trims to control the flow in varies energy systems. The flow capacity of these trims is one of the most important parameters that dictate the effectiveness of the trims, and hence, the effectiveness of the energy systems. The global flow capacity of a geometrically complex trim has been measured experimentally in the present study. The results indicate that as the valve opening

- 1 position increases, the global flow capacity of the trim increases. This increase in the global
- 2 flow capacity of the trim is due to the fact that for the same differential pressure across the
- 3 control valve system, the mass flow rate of the fluid decreases, increasing the flow capacity.
- 4 The decrease in the mass flow rate is a direct consequence of the decrease in the valve
- 5 opening position, offering more resistance to the flow.
- 6

7 The current study uniquely relates the local flow capacity, and hence, the local flow features, with the global flow capacity of the trim. The predicted results indicate that the local capacity 8 within different flow paths of a particular row of the trim remains the same, while it changes 9 from one row to another. It has been noticed that the 3rd row of the trim demonstrates the 10 lowest flow capacity, but highest pressure drop and flow velocity, resulting in severe losses 11 and erosion within the trim. Furthermore, it has also been shown that different disks of a trim 12 13 have different flow capacities, and as the valve opening position increases, the flow capacity of the trim also increases. Based on the global experimental results, and the local numerical 14 predictions, a prediction model has been developed that inter-relates the geometrical features 15 of a trim to its local flow capacity, as the geometrical features dictate the flow capacity of the 16 trim. This prediction model is expected to be a useful tool for trim designers in order to 17 design more efficient trims. Moreover, this prediction model can be integrated with the 18

- 19 design tool, for the energy systems, for integrated performance estimation.
- 20

34

21 Nomenclature

- 22 Q Volumetric flow rate (m^3/hr)
- 23 D Diameter (m)
- 24 p Static gauge pressure (Pa)
- 25 ρ Density of the fluid (kg/m³)
- 26 ρ_o Operating Density (kg/m³)
- 27 ΔP Differential pressure across the valve (kPa)
- 28 VOP Valve Opening Position (%)
- 29 Cv Flow Capacity $(\sqrt{m^7/kg})$
- 30 N_1 Numerical constant (-)
- 31 F_R Reynolds number factor (-)
- 32 F_P Piping geometry factor (-)
- 33 U Flow velocity magnitude (m/sec)

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